



# Direct mapping from LES resolved scales to filtered-flame generated manifolds using convolutional neural networks

Andréa Seltz, Pascale Domingo, Luc Vervisch, Zacharias Nikolaou

## ► To cite this version:

Andréa Seltz, Pascale Domingo, Luc Vervisch, Zacharias Nikolaou. Direct mapping from LES resolved scales to filtered-flame generated manifolds using convolutional neural networks. Combustion and Flame, 2019, 210, pp.71-82. 10.1016/j.combustflame.2019.08.014 . hal-02313873

**HAL Id: hal-02313873**

**<https://normandie-univ.hal.science/hal-02313873>**

Submitted on 4 Dec 2020

**HAL** is a multi-disciplinary open access archive for the deposit and dissemination of scientific research documents, whether they are published or not. The documents may come from teaching and research institutions in France or abroad, or from public or private research centers.

L'archive ouverte pluridisciplinaire **HAL**, est destinée au dépôt et à la diffusion de documents scientifiques de niveau recherche, publiés ou non, émanant des établissements d'enseignement et de recherche français ou étrangers, des laboratoires publics ou privés.

# Direct mapping from LES resolved scales to filtered-flame generated manifolds using convolutional neural networks

Andrea Seltz<sup>a,b</sup>, Pascale Domingo<sup>a</sup>, Luc Vervisch<sup>a</sup>, Zacharias M. Nikolaou<sup>c</sup>

<sup>a</sup>*CORIA - CNRS, Normandie Université, INSA de Rouen, Technopole du Madrillet, BP 8  
76801 Saint-Etienne-du-Rouvray, France*

<sup>b</sup>*Safran Aircraft Engines, Site de Villaroche, Rond-Point René Ravaud-Réau, 77550 Moissy Cramayel,  
France*

<sup>c</sup>*Defence and Security Research Institute, University of Nicosia, Nicosia, Cyprus.*

---

## Abstract

A unified modelling framework for all unresolved terms in the filtered progress variable transport equation in large-eddy simulations of turbulent premixed flames is proposed, using convolutional neural networks. A direct numerical simulation database of a turbulent premixed stoichiometric methane/air jet flame is used in order to train convolutional neural networks to predict both the filtered progress variable source term and the unresolved scalar transport terms. A single variable readily available from the large-eddy simulation is required in order to calculate all inputs to networks, namely the Favre-filtered progress variable  $\tilde{c}$ . In the context of flame tabulated chemistry (premixed flamelet), the trained networks are shown to produce quantitatively good predictions of all unresolved terms in an apriori study, despite their different nature and irrespective of variations in filter size, without having to resort to solving any additional transport equations. The framework proposed in this study thus opens perspectives for the application of deep learning to the modeling of the non-linear aerothermochemistry equations which involve unresolved source and transport terms.

*Keywords:* Turbulent premixed combustion, Deep learning, Flamelet modelling, Flame tabulated chemistry, Machine learning, Neural networks

---

\*Corresponding author

Email address: [domingo@coria.fr](mailto:domingo@coria.fr) (Pascale Domingo)

## 1. Introduction

In Large Eddy Simulation (LES), only the largest flow scales are resolved, while fluctuations of momentum, species concentration, and temperature below the mesh size are unknown. The effects of these unresolved scales appear as unclosed terms in the spatially-  
5 filtered governing equations and require modelling in order to obtain a closed system of equations [1].

With the rapid increase of computational resources and data storage capabilities in recent years, machine learning methods in general have had a tremendous impact in various fields such as speech, image and text recognition, robotics, and health-care amongst many others  
10 [2]. The modelling of unresolved terms in the highly non-linear transport equations of turbulent and reacting flows is a challenging and daunting task. The ability to “learn” from the data directly, presents a promising alternative given the abundance of data available both from simulations and experiments. Direct Numerical Simulation (DNS) databases where all flow and time scales are resolved, are of the order of Petabytes [3] and machine-  
15 learning methods are a natural tool for extracting useful information from these databases for modelling purposes.

In the context of turbulent combustion modelling, Artificial Neural Networks (ANNs) have been used mainly to deal with the introduction of complex chemistry in the simulations [4–8], or to manage complex multi-physics phenomena such as solid-fuel devolatiliza-  
20 tion [9]. Recently, Convolutional Neural Networks (CNNs), originally developed for analysing visual representations [10, 11], have been introduced as a tool for the direct deconvolution of the filtered progress variable [12], which combined with explicit filtering allowed the modelling of the unresolved variance, a key parameter in flamelet modeling [13]. CNNs were also used for modelling the unresolved flame surface wrinkling in [14] surpassing state of the art  
25 explicit algebraic models. CNNs have also been used to extract the chemical rate constant from shock-tube measurements [15] and for predicting the combustion activation energy [16].

A turbulent premixed stoichiometric methane/air jet flame is considered in this study

performing a priori evaluation of neural network based modeling from a fully resolved simulation, following a strategy combining CNNs with the pioneering works of Bray and co-workers [13, 17]. According to their analysis, the departure between the non-linear chemical sources as computed from the node values resolved on a coarse mesh (*i.e.*, neglecting unresolved fluctuations) and their space-filtered (or averaged) counterparts (*i.e.*, accounting for unresolved fluctuations), evolves with the local three-dimensional flame topology, convoluted with the level of mesh resolution, which controls the amplitude of the unresolved fluctuations of temperature and species.<sup>1</sup> Along these lines, we propose to explore the relationships between the three-dimensional distributions of chemical sources as computed from node values (thus a crude approximation of the filtered burning rates), and the filtered value of the non-linear source located at the center of this three-dimensional distribution, using a DNS database. The same procedure is adopted for the sum of the divergence of the unresolved part of the convective flux and of the molecular diffusive flux. The DNS database is then used to train convolutional networks in order to directly reconstruct the unresolved scalar sources and transport terms in the framework of tabulated detailed chemistry (premixed flamelet) LES. The major advantage of such a direct reconstruction of unresolved sources and fluxes from mesh-resolved quantities in the LES, is that by doing so there is no need for explicit filtering or solving additional transport equations, both of which save computational time and mitigate possible resolution issues [18].

## 2. Background and methodology

In chemistry tabulation based on premixed flame generated manifolds, all thermochemical quantities  $\phi$  are uniquely related to an appropriate progress variable  $c$  (equal to unity in fully burnt products and vanishing in fresh gases), so that knowledge of the progress variable distribution  $c(\underline{x}, t)$  is sufficient to characterise the reaction zones, *i.e.*  $\phi(\underline{x}, t) = \phi(c(\underline{x}, t))$  [19–

---

<sup>1</sup>Bray and co-workers specifically focussed on the asymptotic limit where the mesh size is very large compared to the characteristic flame thickness.

22], including the burning rate

$$\dot{\omega}(\underline{x}, t) = \dot{\omega}(c(\underline{x}, t)) . \quad (1)$$

The progress variable may be defined from a set of species mass fractions, temperature, derived from optimisation [23–25] or following other strategies [26]. In any case,  $c$  should  
 55 be a monotonic function through the laminar flamelet. In the context of LES, a transport equation for  $\bar{\rho}\tilde{c}$  is solved,

$$\frac{\partial \bar{\rho}\tilde{c}}{\partial t} + \nabla \cdot (\bar{\rho}\tilde{\mathbf{u}}\tilde{c}) = \nabla \cdot (\bar{\rho}D_c(\tilde{c})\nabla\tilde{c}) + \nabla \cdot \tau + \bar{\dot{\omega}} , \quad (2)$$

where  $\rho$  is the density,  $\mathbf{u}$  is the velocity vector and  $D_c(c)$  is the tabulated molecular diffusion coefficient of  $c$ , defined from the diffusion velocity of tabulated species (Eq. (15) in [27]). The notation  $D_c(\tilde{c})$  means that the diffusion coefficient is here computed from the resolved  
 60 filtered progress variable.  $\bar{\dot{\omega}}(\underline{x}, t)$  is the filtered burning rate of  $c$ . The sub-grid scale flux is

$$\tau = \tau_{D_c} - \tau_c , \quad (3)$$

where  $\tau_{D_c}$  and  $\tau_c$  are respectively the transport of  $c$  by unresolved fluctuations of molecular diffusive flux and momentum,

$$\tau_D = \overline{\rho D_c(c) \nabla c} - \bar{\rho} D_c(\tilde{c}) \nabla \tilde{c} , \quad (4)$$

$$\tau_c = \overline{\rho \mathbf{u} c} - \bar{\rho} \tilde{\mathbf{u}} \tilde{c} . \quad (5)$$

Numerous modeling strategies have been proposed in the literature for the unresolved terms of Eq. (2) in the context of tabulated chemistry, and a detailed review is given in [28]. Among  
 65 those, many involve gradient transport models with an eddy viscosity hypothesis to close  $\tau_c$  while  $\tau_D$  is usually neglected.

Flamelet models for the burning rate  $\bar{\dot{\omega}}$ , are typically based on solving an additional balance equation for the variance of  $c$ ,  $c_v = \tilde{c}^2 - \tilde{c}\tilde{c}$ . A function is then presumed for the

progress variable probability density function (pdf), which is parameterised using the two  
70 moments of  $c$  namely  $\tilde{c}(\underline{x}, t)$  and  $c_v(\underline{x}, t)$  [29–31]. A chemical lookup table is constructed  
using results from 1D flame simulations, and variables of interest are obtained using these  
two parameters from the table. For example, the filtered burning rate in Eq. (2) is closed  
using

$$\bar{\omega}(\underline{x}, t) = \int_0^1 \dot{\omega}(c^*) \bar{P}(c^*; \tilde{c}(\underline{x}, t), c_v(\underline{x}, t)) dc^*, \quad (6)$$

where  $\bar{P}(c^*; \tilde{c}(\underline{x}, t), c_v(\underline{x}, t))$  is the presumed pdf. A characteristic length scale may also be  
75 added to the modeling framework, by combining the pdf with the Flame Surface Density  
(FSD) concept [32, 33]. In an attempt to account for the time history of micro-mixing, it  
has been proposed in [34] to include as a control parameter of the filtered thermo-chemistry  
lookup table, the age of fluid particles since their injection. Simulations coupling flame-  
generated manifolds with pdf transport using Eulerian stochastic fields have also been re-  
80 ported [35, 36]. The filtering of the tabulated one-dimensional flames is another option,  
providing closed expressions for  $\tau_c$  and  $\tau_D$ , in addition to  $\bar{\omega}$  [27, 37, 38]. More recently,  
deconvolution-based approaches have also been discussed and applied to the three terms  $\bar{\omega}$ ,  
 $\tau_c$  and  $\tau_D$  [39–43].

Overall, these modeling approaches directly or indirectly relate  $\tilde{c}$  and  $\nabla \tilde{c}$  to the unclosed  
85 terms. A slightly different approach is explored in this work. First, the statistical properties  
of  $\bar{\omega}$ ,  $\tau_c$  and  $\tau_D$  are examined using the results from the DNS database. Specific features  
are observed in the data connecting  $\bar{\omega}$  and  $\nabla \cdot \tau = \nabla \cdot (\tau_D - \tau_c)$  to  $\dot{\omega}(\tilde{c})$  and  $\nabla \cdot (\bar{\rho} D_c(\tilde{c}) \nabla \tilde{c})$   
respectively, namely the burning rate and the divergence of the diffusive flux as computed  
from the resolved LES fields, *i.e.*, the node values over the LES mesh. ( $\nabla \cdot (\bar{\rho} D_c(\tilde{c}) \nabla \tilde{c})$  is  
90 already calculated when solving for  $\tilde{c}$  and is thus available without additional computational  
cost.) These features suggest that image-type deep learning can be readily applied to dy-  
namically determine two mapping functions  $\mathcal{G}$  and  $\mathcal{F}$  from convolutional neural networks

such that,

$$\bar{\omega}(\underline{x}, t) = \mathcal{G}[\dot{\omega}(\tilde{c}(\underline{x}_1, t)), \dots, \dot{\omega}(\tilde{c}(\underline{x}_N, t))] , \quad (7)$$

$$\nabla \cdot \tau(\underline{x}, t) = \mathcal{F}[\nabla \cdot (\bar{\rho} D_c(\tilde{c}) \nabla \tilde{c})(\underline{x}_1, t), \dots, \nabla \cdot (\bar{\rho} D_c(\tilde{c}) \nabla \tilde{c})(\underline{x}_N, t)] , \quad (8)$$

where  $\tilde{c}(\underline{x}_j, t)$  is known from the LES, with  $\underline{x}_j$  the  $N$  points selected around  $\underline{x}$  to build the  
 95 input image of the networks. Note that the above relations are expressed in progress variable  
 space *i.e.*, a single variable,  $\tilde{c}$ , is required in order to calculate the terms on the right-hand  
 side of Eqs. (7) and (8) which constitute the inputs to the two networks  $\mathcal{F}$  and  $\mathcal{G}$ . Provided  
 $\mathcal{G}$  and  $\mathcal{F}$  are known, Eq. (2) is fully closed without having to solve any additional transport  
 equations. Also note that  $\tilde{c}$  is a coordinate in which turbulent premixed flame properties  
 100 are strongly depended on and feature a generic character when studied in c-space [13]. As a  
 result, the dependence of relations (7) and (8) to the flow regime are expected to be weak as  
 long as the networks are trained for conditions in a Borghi regime-diagram [44] close to the  
 ones of the flames subsequently addressed by LES. This is also more likely to be the case,  
 when  $\mathcal{G}$  and  $\mathcal{F}$  are determined from a reference turbulent premixed flame featuring a large  
 105 degree of flame wrinkling, as is the case for the turbulent premixed jet-flame DNS database  
 used in this study [41].

### 3. Direct simulation database

A previously developed methane-air stoichiometric premixed jet-flame DNS database [41,  
 45, 46] is used for training the neural networks. The configuration is shown in Fig. 1.  
 110 The DNS database is obtained downstream of a well-resolved LES of a piloted premixed  
 stoichiometric fuel-air jet, which generates turbulent flame conditions for the DNS inlet plane  
 located 4.5 diameters downstream of injection. The LES and the DNS are run simultaneously  
 and this is achieved by embedding, inside the LES mesh, a zone where the resolution is  
 sufficiently high so as to resolve the thin reaction zones and the Kolmogorov length scale.  
 115 The configuration is inspired from the experiment by Chen *et al.* [47]. This turbulent Bunsen

burner has a nozzle diameter of  $D = 12$  mm, the jet Reynolds number is 24,000 (bulk nozzle velocity of  $30 \text{ m} \cdot \text{s}^{-1}$  and turbulent kinetic energy of  $3.82 \text{ m}^2 \cdot \text{s}^{-2}$ ). The pilot is set to fully burnt gases at  $T_b = 2200$  K. The LES mesh consists of about 171 million nodes covering a domain  $16D \times 8D \times 8D$ , with a resolution of the order of  $150 \text{ } \mu\text{m}$  (Fig. 1). The resolution in the DNS zone is fixed at  $50 \text{ } \mu\text{m}$ , which was calibrated to ensure a full resolution of the flow and flame scales for this jet flame having a Karlovitz number varying between 1 and 3 [41, 47]. Chemistry tabulation with a stoichiometric premixed flamelet with fresh gases at  $T_o = 300$  K (GRI-3.0 mechanism [48] and progress variable defined from CO, CO<sub>2</sub>, H<sub>2</sub>O and NO<sub>x</sub> as in [49]) is used for both LES and DNS, without any SGS modeling in the DNS part (SGS terms set to zero). The flame thermal thickness based on the progress variable field is of the order of  $\delta_L \approx 400 \text{ } \mu\text{m}$ . The DNS zone consists of 28.58 million nodes ( $243 \times 343 \times 343$ ), over a physical domain of  $12 \text{ mm} \times 18 \text{ mm} \times 18 \text{ mm}$ . This DNS zone is located at  $4.5D$  downstream of the nozzle, and at  $5.5D$  the mesh is coarsened again to progressively resume the simulation using LES (Fig. 1). A progress variable presumed pdf approach is applied in the LES zones [29] and the SGS momentum fluxes are approximated with the Vreman model [50]. These simulations have been performed using the flow solver SiTCom [51], which solves the Navier-Stokes equations in their fully compressible form together with the balance equation for the filtered progress variable. The convective terms are discretised with a fourth-order centered skew-symmetric-like scheme [52] and the diffusive terms with a fourth-order centered scheme. Time is advanced explicitly with a third order Runge-Kutta method and NSCBC boundary conditions [53] are imposed at inlet and outlet, with the measured profiles with synthetic turbulence [54] prescribed at inlet. More details on the development and the use of this DNS database may be found in [41, 45, 46].

A Gaussian filtering operation,  $G(\underline{x}) = (6/(\pi\Delta^2))^{3/2} \exp(-6\underline{x} \cdot \underline{x}/\Delta^2)$ , with filter size  $\Delta = 0.3 \text{ mm} = 0.75 \delta_L$ ,  $\Delta = 0.6 \text{ mm} = 1.50 \delta_L$  and  $\Delta = 0.9 \text{ mm} = 2.25 \delta_L$ , is applied to the DNS variables in order to generate a priori LES filtered quantities, thus varying the resolution of the a priori fields from well-resolved to coarse LES (at least from the reaction



zone point of view,  $\Delta = 0.9$  mm is 18 times larger than the DNS grid resolution).

## 4. Statistical analysis of unresolved terms

### 4.1. Turbulent flame properties

Figure 2 shows  $\langle \overline{\dot{\omega}^+} | \tilde{c} \rangle$ , the statistical mean over the DNS domain of the normalised filtered progress variable source, conditioned on values of  $\tilde{c}$ . The subscript ‘+’ denotes source terms normalised by their maximum value in the tabulated freely-propagating laminar premixed flame. The result obtained using a 1D laminar flame,  $\dot{\omega}^+(\tilde{c})$ , is shown as a solid line. As expected, the maximum of  $\langle \overline{\dot{\omega}^+} | \tilde{c} \rangle$  decreases with increasing filter size and thus with increasing unresolved fluctuations [13]. Following the thickening of the filtered flame front, the response of this conditional filtered source term also spreads in progress variable space for increasing filter sizes, up to  $\Delta = 0.9$  mm.

The statistical conditional means of  $\langle \nabla \cdot \tau_c | \tilde{c} \rangle$  and of  $\langle \nabla \cdot \tau_D | \tilde{c} \rangle$ , the divergence of the convective and diffusive fluxes (Eqs. (4) and (5)), are shown in Fig. 3. The maximum level of velocity fluctuations observed in the fresh gases in the experiment at the streamwise location of the jet where the DNS zone is located (Fig. 1), is of the order of  $u' = 1.80 \text{ m} \cdot \text{s}^{-1}$  [47]. Then, the ratio  $u'/S_L$  for this stoichiometric premixed methane-air flame is of the order of 5, with  $S_L = 0.37 \text{ m} \cdot \text{s}^{-1}$ .

The number  $N_B = [(T_b - T_o)/T_o]S_L/(2\alpha u')$ , as defined by Veynante *et al.* [55], which differentiates between gradient transport,  $N_B < 1$ ,  $-\tau_c \propto \nabla \tilde{c}$ , and counter-gradient transport,  $N_B > 1$ ,  $-\tau_c \propto -\nabla \tilde{c}$ , in a Reynolds Averaged Navier Stokes context (RANS), is above unity in the present case for an efficiency factor  $\alpha \leq 0.6$ . The factor  $\alpha$  in  $N_B$  accounts for the variability in the capability of turbulent eddies to wrinkle the reaction zone [55].

Considering space-filtered (LES) quantities, for  $1 \leq \Delta/\delta_L \leq 3$ , counter-gradient SGS transport was recently reported from DNS analysis for the same level of  $u'/S_L$  [43]. Overall, counter-gradient transport is found when  $\nabla \cdot \tau_c$  and  $\nabla^2 \tilde{c}$  are of same sign. This is also what is observed in Fig. 3(a), with  $\langle \nabla \cdot \tau_c | \tilde{c} \rangle$  negative on the burnt gas side where  $\nabla^2 \tilde{c} < 0$  and

$\langle \nabla \cdot \tau_c \mid \tilde{c} \rangle$  positive on the fresh side where  $\nabla^2 \tilde{c} > 0$ .<sup>2</sup> This behaviour is also observed in the  
 170 scatter plot of the SGS convection divergence which is shown in Fig. 4, with the occurrence  
 however of some negative values of  $\nabla \cdot \tau_c$  around  $\tilde{c} \rightarrow 0$ , thus gradient transport in the  
 preheat zone ensures the local flame propagation. On these scatter plots, the bounds of  
 $\nabla \cdot \tau_c$  do not change much with the filter size  $\Delta$ , but the spreading of the data for a given  
 value of  $\tilde{c}$  decreases with  $\Delta$ .

175 The contribution of the SGS diffusive flux,  $\nabla \cdot \tau_D$ , in Fig. 3(b) cannot be neglected  
 compared to the convective one,  $\nabla \cdot \tau_c$ , in Fig. 3(a). This would not be the case in the  
 RANS context, where the SGS diffusive contribution would be inversely proportional to the  
 turbulent Reynolds number of the flow, and thus could be neglected when compared to other  
 transport terms [56]. The SGS diffusive fluxes in LES are actually inversely proportional  
 180 to the turbulent Reynolds numbers of the LES mesh cells, based on the filter size and on  
 the SGS velocity fluctuations. Therefore, the SGS turbulent Reynolds number appears too  
 small for neglecting the divergence of  $\tau_D$ . The response of the amplitude of  $\langle \nabla \cdot \tau_D \mid \tilde{c} \rangle$  versus  
 the filter size, is better understood by looking at the two terms  $\left\langle \overline{\nabla \cdot (\rho D_c(c) \nabla c)} \mid \tilde{c} \right\rangle$  and  
 $\langle \nabla \cdot (\bar{\rho} D_c(\tilde{c}) \nabla \tilde{c}) \mid \tilde{c} \rangle$  in Figs. 3(c) and 3(d). As expected, following the decay of the gradients  
 185 with the increase of the filter size, these filtered transport terms decrease, leading to a decay  
 of the amplitude of both the filtered and node-resolved diffusive budgets when  $\Delta$  increases.  
 In the case of  $\langle \nabla \cdot (\bar{\rho} D_c(\tilde{c}) \nabla \tilde{c}) \mid \tilde{c} \rangle$  an almost self-similar behaviour is observed against  $\Delta$   
 (Fig. 3(d)). This is not the case for  $\left\langle \overline{\nabla \cdot (\rho D_c(c) \nabla c)} \mid \tilde{c} \right\rangle$ , for which the response is also  
 shifted against  $\tilde{c}$  when  $\Delta$  varies as one may observe from the results in Fig. 3(c). As a result,  
 190 the difference between these two terms,  $\nabla \cdot \tau_D$ , is not monotonic against  $\Delta$  (Fig. 3(b)). It  
 is important also to note that the thickening of the flame front in physical space resulting  
 from filtering, directly impacts these budgets here visualised in  $\tilde{c}$ -space. Finally, the sum  
 of SGS fluxes  $\langle \nabla \cdot \tau \mid \tilde{c} \rangle = \langle \nabla \cdot (\tau_D - \tau_c) \mid \tilde{c} \rangle$ , which combines responses of both unresolved

---

<sup>2</sup>Almost zero fluxes on the burnt gas side appear before  $\tilde{c} = 1$ , because of the choice of the progress  
 variable as in Godel et al. [49], which is designed as slowly varying approaching burnt gases to preserve a  
 single-valued response of NOx versus progress variable.

convection and molecular diffusion, is shown in Fig. 5.

#### 195 4.2. Physical arguments for CNN training

As will be explained later in the text, two CNN will be used to approximate respectively the values of  $\nabla \cdot \tau(\underline{x}, t)$  and  $\bar{\omega}(\underline{x}, t)$  from an input composed of a set of data (images). In practice, this is done by interpolating over a large number of relationships between ‘images’ and ‘labels’, which were the values of  $\nabla \cdot \tau(\underline{x}, t)$  and  $\bar{\omega}(\underline{x}, t)$  “learned” during a training  
 200 phase. Here the inputs (images) are composed of  $\dot{\omega}(\tilde{c}(\underline{x}_j, t))$  and  $\nabla \cdot (\bar{\rho} D_c(\tilde{c}) \nabla \tilde{c})(\underline{x}_j, t)$  for  $j = 1, \dots, N$ , where  $N$  is the number of points surrounding a point  $\underline{x}$ , where the values of  $\bar{\omega}(\underline{x}, t)$  and  $\nabla \cdot \tau(\underline{x}, t)$  are sought.

The input set of data should feature specific topological properties, which can be extracted by convoluting the data points with a series of filters and specific operations. Figure 6 shows  $\langle \bar{\omega}^+ | \tilde{c} \rangle$  versus  $\dot{\omega}^+(\tilde{c})$  for different filter sizes and Fig. 7 illustrates the image-label  
 205 relationship which could be implemented. Notice that the CNN will not operate on the statistical conditional means in the end, but directly on the raw data, however initiating the analysis at the statistical level helps to select the variables. As one may observe from Fig. 6, the filtered source term is not a single-valued function of the node resolved source, nevertheless accounting also for the local curvature of the data set as one of the features, should be  
 210 sufficient to build a one-to-one response. Obviously, this constitutes only a very preliminary condition to secure the determination of the function  $\mathcal{G}$  of Eq. (7) and more features will need to be extracted on the full set of turbulent data, as discussed thereafter. Because of non-deterministic local sub-grid scale wrinkling of the flame surface, the relation between  
 215  $\dot{\omega}^+(\tilde{c})$  and  $\bar{\omega}^+$  is actually scattered as shown in Fig. 8, with some deviation from the response seen in Fig. 6, and this scattering should be reproduced by a reliable physical model. This is where numerical modelling can take great benefit from deep learning, which automatically discovers the most relevant signal features through elementary operations, to then allow for interpolating over the very large dataset learned.

220 Similarly, Fig. 9 shows  $\langle \nabla \cdot \tau | \tilde{c} \rangle$  versus  $\langle \nabla \cdot (\bar{\rho} D_c(\tilde{c}) \nabla \tilde{c}) | \tilde{c} \rangle$  and Fig. 10 the full set of

data  $\nabla \cdot \tau$  versus  $\nabla \cdot (\bar{\rho} D_c(\tilde{c}) \nabla \tilde{c})$ , revealing a dataset which can easily be analysed by CNNs for identifying the function  $\mathcal{F}$  in Eq. (8). It will be seen thereafter that this is a valid option for the case considered. Because momentum also contributes to  $\nabla \cdot \tau$ , an alternative would consist of introducing information on velocity in the  $\mathcal{F}$  neural network (Eq. (8)). An option that was not found necessary in the present study where both SGS convective and diffusive fluxes are combined to build a single CNN for the divergence of fluxes. For this set of data, various options in terms of neural network layers number and filtering kernels have been tried. Best results were obtained with two layers and the set of kernels and data organisation now reported.

## 5. CNN training process

The LES mesh size required to resolve with  $n = 5$  points the filtered progress variable signal, may be estimated from  $h = (\Delta/n) \sqrt{\pi/6 + \delta_L^2/\Delta^2}$  [42]. A three-dimensional test-box of size  $(2h)^3$  is constructed around every of the  $M = 28.58$  million DNS nodes. This test box is centered at  $\underline{x}$  and contains  $N = 27$  points which hold the three-dimensional distributions of  $\dot{\omega}(\tilde{c}(\underline{x}_j, t))$  and  $\nabla \cdot (\bar{\rho} D_c(\tilde{c}) \nabla \tilde{c})(\underline{x}_j, t)$ , for  $j = 1, \dots, N$ . These data are stored and constitute the ‘images’ that will be processed by the CNN as shown in Fig. 11. The ‘labels’ of each  $i$ -th image are  $\bar{\omega}[i] = \bar{\omega}(\underline{x}, t)$  and  $\nabla \cdot \tau[i] = \nabla \cdot (\tau_D(\underline{x}, t) - \tau_c(\underline{x}, t))$  for  $i = 1, \dots, N_L$ . Two networks of similar structures (same number of layers, convolution kernels, etc.) are trained, one for the chemical source and one for the SGS fluxes.

To reduce the computational cost, only part of the database is used for training. For each value of  $\Delta$ , the following procedure is applied:

- First, 1000 images with their associated  $i$ -th label are built. 20 values of  $c^*$  uniformly distributed between 0 and 1 ( $\Delta c^* = 0.05$ ) are defined. For each value of  $c^*$ , 50 images are randomly selected so that  $\tilde{c}(\underline{x}, t) \in [c^* - \Delta c^*/2; c^* + \Delta c^*/2]$  ( $\underline{x}$  denotes the center of the test box, Fig. 11).

- Overfitting is avoided by adding uncorrelated random perturbations to the images,  $\dot{\omega}(\tilde{c}(\underline{x}_j, t))$  and  $\nabla \cdot (\bar{\rho} D_c(\tilde{c}) \nabla \tilde{c})(\underline{x}_j, t)$ , as 10% of their maximum in the test box, to build a second image for each label. 2000 images are then available for 1000 labels.

Finally, the database used for training with two filter sizes contains 4000 images and  $N_L =$   
250 2000 labels. Hence, for every quantity studied, a set of  $27 \times 4000 = 108000$  data ( $\dot{\omega}(\tilde{c}(\underline{x}_j, t))$   
and  $\nabla \cdot (\bar{\rho} D_c(\tilde{c}) \nabla \tilde{c})(\underline{x}_j, t)$ ) is involved, associated to the 2000 reference labels ( $\bar{\omega}[i]$  and  
 $\nabla \cdot \tau[i]$  for  $i = 1, \dots, N_L$ ).

A series of convolution/sampling operations are done iteratively during the training  
phase, in which the neural weights are adjusted until a satisfying minimal error is obtained  
255 between the value of  $\nabla \cdot \tau(\underline{x}, t)$  and  $\bar{\omega}(\underline{x}, t)$  used for training (labels) and the values returned  
by the neural networks prediction. Convolution/sampling operations are thus performed on  
the database to extract its features using a number of different kernels [10, 11]:

1. Each image is convoluted with 32 different filter kernels obtained with random values  
from a truncated normal distribution. Meaningful values of the obtained features are  
260 then extracted with a max pooling non-linear function to avoid excessive computational  
costs.
2. The process is repeated with 64 filters, decomposing the image into several meaningful  
features, which is useful for seeking out the inner properties of the fluxes and sources.
3. Two fully connected layers are built to process the 64 obtained features, and to classify  
265 the image according to the learned labels. The probabilities linking this image to each  
of the learned labels are then known in the form of coefficients ranging between zero  
and unity.

The training of the network was conducted using the TensorFlow ([www.tensorflow.org](http://www.tensorflow.org)) li-  
brary and breakdown of the network structure is given in Fig. 12. During this training phase,  
270 a 50% drop-out rate is applied, *i.e.*, 2000 images are randomly selected at every iteration  
and about 100 iterations (or ‘epoch’) are needed to reach convergence. The error function  
used for training is based on cross entropy [57], while the training is controlled by the Adam

optimizer [58] for stochastic gradient descent, with a user-defined learning rate of  $10^{-4}$ . Both normalised and non-normalised input ( $\dot{\omega}(\tilde{c})$  and  $\nabla \cdot (\bar{\rho} D_c(\tilde{c}) \nabla \tilde{c})$ ) and output values ( $\bar{\dot{\omega}}$  and  $\nabla \cdot \tau$ ) of networks have been used, without much difference, results are presented for the non-normalised training.

## 6. CNN mapping of fluxes and sources from LES resolved fields

In using the networks, the  $N = 27$  values of the chemical sources and of the divergence of the fluxes computed from the resolved progress variable field in the test box surrounding the LES cell (Fig. 11), constitute the input. In this feasibility study, for each filter size, 1000 filtered DNS fields are used for a priori tests (the noised images introduced during the training phase do not enter these tests). The unknown terms are then approximated from interpolation over the  $N_L = 2000$  labels values ( $\bar{\dot{\omega}}[i]$  and  $\nabla \cdot \tau[i]$ ) of the training phase,

$$\begin{aligned} \bar{\dot{\omega}}(\underline{x}, t) &= \mathcal{G}[\dot{\omega}(\tilde{c}(\underline{x}_1, t)), \dots, \dot{\omega}(\tilde{c}(\underline{x}_N, t))] \\ &= \sum_{i=1}^{N_L} P_i(\underline{x}, t) \times \bar{\dot{\omega}}[i], \end{aligned} \quad (9)$$

$$\begin{aligned} \nabla \cdot \tau(\underline{x}, t) &= \mathcal{F}[\nabla \cdot (\bar{\rho} D_c(\tilde{c}) \nabla \tilde{c}(\underline{x}_1, t)), \dots, \nabla \cdot (\bar{\rho} D_c(\tilde{c}) \nabla \tilde{c}(\underline{x}_N, t))] \\ &= \sum_{i=1}^{N_L} T_i(\underline{x}, t) \times \nabla \cdot \tau[i], \end{aligned} \quad (10)$$

where  $P_i(\underline{x}, t)$  and  $T_i(\underline{x}, t)$ , both  $\in [0, 1]$ , are the probability that the image belongs to the label ‘ $i$ ’, as returned by the neural networks  $\mathcal{G}$  and  $\mathcal{F}$ . In practice, the modelled filtered sources and divergence of SGS fluxes are thus non-linearly interpolated, according to the local LES resolved flame topology, over 2000 reference DNS values.

The training is performed for the smaller and largest filter sizes *i.e.*,  $\Delta = 0.3$  mm and  $\Delta = 0.9$  mm. Then, the prediction capabilities of the obtained CNN are tested a priori for these filter sizes and for intermediate values of  $\Delta \in [0.3, 0.9]$ , for which this network has not been trained (so-called ‘untrained case’). Notice that the ratio of three between the filter sizes used for training can be considered large, as these filters vary between  $0.75 \delta_L$  and

2.25  $\delta_L$ . This ratio of more than one flame thickness is here intentional to test the method in the limit case where the neural networks are used for filter sizes far from those of their training.

Using GPU ‘NVIDIA Pascal’, the training of the fluxes requires 4 hours. Compared to the fluxes, the filtered source terms have a larger range of variation between the two filter sizes used for training and they require 24 hours of training on the same GPU. Once trained, the network may be used directly in a flow-solver for a CPU cost of about the one required with a turbulent combustion closure based on chemistry tabulation and presumed probability density function [29].

The averages of the predicted divergence of the unresolved fluxes conditioned on the progress variable,  $\langle \nabla \cdot \tau \mid \tilde{c} \rangle$ , are first compared against the filtered DNS in Fig. 13. The CNN reproduces the expected behaviour and amplitude of  $\nabla \cdot \tau$ , the fluctuations are also well captured, as seen in Fig 14. (Note that because the binning intervals to compute conditional means are different than in Fig. 5, the extrema also differ.) The test for the untrained filter level is performed with a filter size  $\Delta = 0.45 \text{ mm} = 1.125\delta_L$ . This constitutes a stringent test case, because neural networks are known to be prone to rapid divergence when applied away from their training area. However, staying within the bounds of the training filter sizes, Fig. 13(b) shows that the response of the divergence of the unresolved fluxes is well captured. The plots showing conditional fluctuations in Fig. 14(b) confirm this moderate deviation from the reference filtered DNS. These results need to be put in perspective with predictions of unresolved fluxes using most advanced SGS models, where sometimes even the sign is not properly returned (see for instance Fig. 8 of [43] reporting strong departure from DNS in SGS transport modeling in turbulent premixed flames).

Similar results are obtained for the filtered source terms, which are shown in Figs. 15 and 16. For the trained filter sizes ( $\Delta = 0.3$  and  $0.9 \text{ mm}$ ), the filtered chemical source as predicted by the CNN matches the DNS reference, specifically for the largest filter, with a good reproduction of the parabolic shape. The conditional fluctuations of filtered burning

320 rates are also well captured (Fig. 16). For the untrained case however, some departure  
 from the DNS value is observed, but still reasonable, at least comparable to what could be  
 expected using classic models to estimate the filtered source terms. This would particularly  
 be the case against formulations where the Arrhenius form is kept at the resolved scales  
 after simply applying a scaling factor, thus far from the parabolic shape developing with the  
 325 increase in filter size. In a previous work [41], modeling of the filtered source based on 3D  
 approximate deconvolution and 1D flame deconvolution was tested against the same DNS  
 database. As shown in Fig.17(b) of [41], the error on the burning rate estimation conditioned  
 on the progress variable could reached up to 25% for  $\Delta = 3\delta_L$ . In the present case with the  
 neural network, the maximum error is of the order of 1% on the trained database and of  
 330 16% for the untrained ones, confirming the potential of the approach.

## 7. Conclusion

A novel modelling framework using machine-learning is proposed for providing closures  
 for all unresolved terms in the filtered transport equation of the progress variable in large-  
 eddy simulations of turbulent premixed flames in the context of flamelet tabulated chemistry.  
 335 A turbulent premixed methane/air stoichiometric premixed jet flame is considered and a  
 priori evaluation of modeling based on neural networks is performed.

Convolutional neural networks are trained using data from a direct numerical simulation  
 database, in order to predict the filtered progress-variable source term, and the unresolved  
 fluxes in the filtered transport equation of  $\tilde{c}$ . The advantage of the approach proposed in this  
 340 study, is that a single variable which is readily available,  $\tilde{c}$ , is required in order to calculate all  
 inputs to networks, without having to resort to solving any additional transport equations for  
 modelling all terms in the transport equation. The convolutional neural networks are shown  
 to provide quantitatively accurate predictions of both the source and flux terms, which  
 are two substantially different terms and otherwise difficult to model in a single unified  
 345 framework. The predictions capabilities of the networks are also demonstrated to be only



weakly insensitive to variations in filter width, which is an important attribute for any sub-grid scale model. Because they are based on the progress variable, a generic parameter of premixed flames, the networks should perform well for any turbulent premixed flame located in the Borghi regime-diagram close to the conditions used for training. However, as  
350 a non-linear interpolation tool of high-dimensionality, it cannot perform well for cases with operating conditions far from those of training.

## Acknowledgment

The PhD of the first author is funded by ANRT (Agence Nationale de la Recherche et de la Technology), SAFRAN-AE under the CIFRE No 1643/2016. Computing time has  
355 been provided by CRIANN (Centre Régional Informatique et d’Applications Numériques de Normandie). This work was granted access to the HPC resources of IDRIS-CNRS under the allocation 2018-A0052B00152 made by GENCI (Grand Equipement National de Calcul Intensif).

## References

- 360 [1] H. Pitsch, Large Eddy Simulation of Turbulent Combustion, Annual Review of Fluid Mechanics 38 (2006) 453–482.
- [2] R. Wason, Deep learning: Evolution and expansion, Cognitive Systems Research 52 (2018) 701–708.
- [3] K. Kanov, R. Burns, C. Lalescu, G. Eyink, The Johns Hopkins turbulence databases:  
365 an open simulation laboratory for turbulence research, Comput. Sci. Eng. 17 (2015) 10–17.
- [4] J. A. Blasco, N. Fueyo, C. Dopazo, J. Ballester, Modelling the temporal evolution of a reduced combustion chemical system with artificial neural network, Combust. Flame 113 (1-2) (1998) 38–52.

- [5] F. C. Christo, A. R. Masri, E. M. Nebot, Artificial neural network implementation of chemistry with pdf simulation of H<sub>2</sub>/CO<sub>2</sub> flames, *Combust. Flame* 106 (1996) 406–427.
- [6] B. A. Sen, E. R. Hawkes, S. Menon, Large eddy simulation of extinction and reignition with artificial neural networks based chemical kinetics, *Combust. Flame* 157 (3) (2010) 566–578.
- [7] B. A. Sen, S. Menon, Large eddy mixing based tabulation and artificial neural networks for large eddy simulations of turbulent flames, *Combust. Flame* 157 (1) (2010) 62–74.
- [8] L. L. Franke, A. K. Chatzopoulos, S. Rigopoulos, Tabulation of combustion chemistry via Artificial Neural Networks (ANNs): Methodology and application to LES-PDF simulation of Sydney flame L, *Combust. Flame* 185 (2017) 245–260.
- [9] T. Abbas, M. M. Awai, F. C. Lockwood, An artificial intelligence treatment of devolatilization for pulverized coal and biomass in co-fired flames, *Combust. Flame* 132 (3) (2003) 305–318.
- [10] Y. Lecun, Y. Bengio, G. Hinton, Deep learning, *Nature* 521 (2015) 436–444.
- [11] J. Schmidhuber, Deep learning in neural networks: An overview, *Neural Networks* 61 (2015) 85–117.
- [12] Z. Nikolaou, C. Chrysostomou, L. Vervisch, S. Cant, Modelling turbulent premixed flames using convolutional neural networks: application to sub-grid scale variance and filtered reaction rate, *Flow Turbulence Combust.* (in press) .
- [13] K. N. C. Bray, The Challenge of turbulent combustion, *Symp. (Int.) on Combust.* 26 (1996) 1–26.
- [14] C. J. Lapeyre, A. Misdariis, N. Cazards, D. Veynante, T. Poinso, Training convolutional neural networks to estimate turbulent sub-grid scale reaction rates, *Combust. Flame* 203 (2019) 255–264.
- [15] R. Ranade, S. Alqahtani, A. Farooq, T. Echehki, An ANN based hybrid chemistry framework for complex fuels, *Fuel* 241 (2019) 625–636.
- [16] W. Jiang, X. Xing, X. Zhang, M. Mi, Prediction of combustion activation energy of

NaOH/KOH catalyzed straw pyrolytic carbon based on machine learning, *Renewable energy* 130 (2019) 1216–1225.

[17] J.-B. Moss, K. Bray, A unified statistical model of the premixed turbulent flame, *Acta Astronautica* 4 (1977) 291–319.

[18] C. Mehl, J. Idier, B. Fiorina, Evaluation of deconvolution modelling applied to numerical combustion, *Combust. Theory Modelling* 22 (1) (2018) 38–70.

[19] O. Gicquel, N. Darabiha, D. Thevenin, Laminar Premixed Hydrogen / Air Counterflow Flame Simulations Using Flame Prolongation of ILDM With Differential Diffusion, *Proc. Comb. Inst.* 28 (2000) 1901–1908.

[20] J. A. van Oijen, F. A. Lammers, L. P. H. de Goey, Modeling of Complex Premixed Burner Systems by Using Flamelet-Generated Manifolds, *Combust. Flame* 127 (3) (2001) 2124–2134.

[21] P.-D. Nguyen, L. Vervisch, V. Subramanian, P. Domingo, Multidimensional flamelet-generated manifolds for partially premixed combustion, *Combust. Flame* 157 (1) (2010) 43–61.

[22] A. Scholtissek, P. Domingo, L. Vervisch, C. Hasse, A self-contained progress variable space solution method for thermochemical variables and flame speed in freely-propagating premixed flamelets, *Proc. Combust. Inst.* 27 (2) (2019) 1529–1536.

[23] Y.-S. Niu, L. Vervisch, P. D. Tao, An optimization-based approach to detailed chemistry tabulation: Automated progress variables definition, *Combust. Flame* 160 (4) (2013) 776–785.

[24] U. Prufert, S. Hartl, F. Hunger, D. Messig, M. Eiermann, C. Hasse, A constrained control approach for the automated choice of an optimal progress variable for chemistry tabulation, *Flow Turbulence Combust.* 94 (3) (2015) 593–617.

[25] M. Ihme, L. Shunn, J. Zhang, Regularization of reaction progress variable for application to flamelet-based combustion models, *J. Comput. Phys.* 231 (2012) 7715–7721.

[26] M.-S. Benzinger, R. Schiessl, U. Maas, A versatile coupled progress variable / REDIM

model for auto-ignition and combustion, *Proc. Combust. Inst.* 36 (3) (2017) 3613–3621.

425 [27] S. Nambully, P. Domingo, V. Moureau, L. Vervisch, A Filtered-Laminar-Flame PDF  
sub-grid scale closure for LES of premixed turbulent flames. Part I: Formalism and  
application to a bluff-body burner with differential diffusion, *Combust. Flame* 161 (7)  
(2014) 1756–1774.

[28] J. A. van Oijen, A. Donini, R. J. M. Bastiaans, J. H. M. tenThije Boonkkamp, L. P. H.  
430 de Goey, State-of-the-art in premixed combustion modeling using flamelet generated  
manifolds, *Progress Energy Combust. Sci.* 57 (2016) 30 – 74.

[29] P. Domingo, L. Vervisch, D. Veynante, Large-Eddy Simulation of a lifted methane-air  
jet flame in a vitiated coflow, *Combust. Flame* 152 (3) (2008) 415–432.

[30] V. Subramanian, P. Domingo, L. Vervisch, Large-eddy simulation of forced ignition of  
435 an annular bluff-body burner, *Combust. Flame* 157 (3) (2010) 579–601.

[31] J. Lai, N. Chakraborty, Modeling of progress variable variance transport in head-on  
quenching of turbulent premixed flames: A direct numerical simulation analysis, *Com-  
bust. Sci. Tech.* 188 (11-12) (2016) 1925–1950.

[32] G. Lecocq, S. Richard, O. Colin, L. Vervisch, Hybrid presumed pdf and flame surface  
440 density approach for Large-Eddy Simulation of premixed turbulent combustion, Part 1:  
Formalism and simulations of a quasi-steady burner, *Combust. Flame* 158 (6) (2011)  
1201–1214.

[33] G. Lecocq, S. Richard, O. Colin, L. Vervisch, Hybrid presumed pdf and flame surface  
density approach for Large-Eddy Simulation of premixed turbulent combustion, Part 2:  
445 Early flame development after sparking, *Combust. Flame* 158 (6) (2011) 1215–1226.

[34] N. Enjalbert, P. Domingo, L. Vervisch, Mixing time-history effects in Large Eddy Simu-  
lation of non-premixed turbulent flames: Flow-Controlled Chemistry Tabulation, *Com-  
bust. Flame* 159 (1) (2012) 336–352.

[35] L. Valino, A field monte carlo formulation for calculating the probability density function  
450 of a single scalar in turbulent flow, *Flow Turbulence Combust.* 60 (2) (1998) 157–172.

- [36] A. Avdić, G. Kuenne, J. Janicka, Flow physics of a bluff-body swirl stabilized flame and their prediction by means of a joint Eulerian stochastic field and tabulated chemistry approach, *Flow Turbulence Combust.* 97 (4) (2016) 1185–1210.
- [37] B. Fiorina, R. Vicquelin, P. Auzillon, N. Darabiha, O. Gicquel, D. Veynante, A filtered  
455 tabulated chemistry model for LES of premixed combustion, *Combust. Flame* 157 (2010) 465–475.
- [38] R. Mercier, T. Schmitt, D. Veynante, B. Fiorina, The influence of combustion SGS submodels on the resolved flame propagation. Application to the LES of the Cambridge stratified flames, *Proc. Combust. Inst.* 35 (2) (2015) 1259–1267.
- 460 [39] P. Domingo, L. Vervisch, Large Eddy Simulation of premixed turbulent combustion using approximate deconvolution and explicit flame filtering, *Proc. Combust. Inst.* 35 (2) (2015) 1349–1357.
- [40] Q. Wang, M. Ihme, Regularized deconvolution method for turbulent combustion modeling, *Combust. Flame* 176 (2017) 125–142.
- 465 [41] P. Domingo, L. Vervisch, DNS and approximate deconvolution as a tool to analyse one-dimensional filtered flame sub-grid scale modeling, *Combust. Flame* 177 (2017) 109–122.
- [42] Z. Nikolaou, L. Vervisch, A priori assessment of an iterative deconvolution method for LES sub-grid scale variance modelling, *Flow Turbulence Combust.* 101 (1) (2018) 33–53.
- 470 [43] Z. Nikolaou, R. S. Cant, L. Vervisch, Scalar flux modelling in turbulent flames using iterative deconvolution, *Phys. Rev. Fluids.* 3 (4) (2018) 043201.
- [44] R. Borghi, Mise au point sur la structure des flammes turbulentes, *J. Chimie Physique* 81 (6) (1984) 361–370.
- [45] L. Cifuentes, C. Dopazo, J. Martin, P. Domingo, L. Vervisch, Local volumetric dilata-  
475 tion rate and scalar geometries in a premixed methane-air turbulent jet flame, *Proc. Combust. Inst.* 35 (2) (2015) 1295–1303.
- [46] L. Cifuentes, C. Dopazo, J. Martin, P. Domingo, L. Vervisch, Effects of the local flow

topologies upon the structure of a premixed methane-air turbulent jet flame, *Flow Turbulence Combust.* 96 (2) (2016) 535–546.

- 480 [47] Y.-C. Chen, N. Peters, G. A. Schneemann, N. Wruck, U. Renz, M. S. Mansour, The detailed flame structure of highly stretched turbulent premixed methane-air flames, *Combust. Flame* 107 (3) (1996) 223–244.
- [48] G. P. Smith, D. M. Golden, M. Frenklach, N. W. Moriarty, B. Eiteneer, M. Goldenberg, C. T. Bowman, R. K. Hanson, S. Song, W. C. Gardiner, V. V. Lissianski, Z. Qin, Tech. Rep., <http://www.me.berkeley.edu/gri-mech/>, 1999.
- 485 [49] G. Godel, P. Domingo, L. Vervisch, Tabulation of NO<sub>x</sub> chemistry for Large-Eddy Simulation of non-premixed turbulent flames, *Proc. Combust. Inst.* 32 (2009) 1555–1561.
- [50] A. W. Vreman, An eddy-viscosity subgrid-scale model for turbulent shear flow: Algebraic theory and applications, *Phys. Fluids*. 16 (10) (2004) 3670–3681.
- 490 [51] L. Bouheraoua, P. Domingo, G. Ribert, Large-eddy simulation of a supersonic lifted jet flame: Analysis of the turbulent flame base, *Combust. Flame* 179 (2017) 199–218.
- [52] F. Ducros, F. Laporte, T. Soulères, V. Guinot, P. Moinat, B. Caruelle, High-order fluxes for conservative skew-symmetric-like schemes in structured meshes: application to compressible flows, *J. Comput. Phys.* 161 (2000) 114–139.
- 495 [53] G. Lodato, P. Domingo, L. Vervisch, Three-dimensional boundary conditions for Direct and Large-Eddy Simulation of compressible viscous flows, *J. Comput. Phys* 227 (10) (2008) 5105–5143.
- [54] M. Klein, A. Sadiki, J. Janicka, A digital filter based generation of inflow data for spatially developing direct numerical or large eddy simulations, *J. Comp. Physics* 186 (2) 500 (2002) 652–665.
- [55] D. Veynante, A. Trouvé, K. Bray, T. Mantel, Gradient and counter-gradient scalar transport in turbulent premixed flames, *J. Fluid Mech.* 332 (1997) 263–293.
- [56] D. Veynante, L. Vervisch, Turbulent Combustion Modeling, *Prog Energy Combust Sci* 28 (2002) 193–266.

- 505 [57] P.-T. de Boer, D. P. Kroese, S. S. Mannor, R. Y. Rubinstein, A tutorial on the cross-entropy method, *Annals Operations Research* 134 (1) (2005) 19–67.
- [58] D. P. Kingma, J. L. Ba, ADAM: A method for stochastic optimization, <https://arxiv.org/pdf/1412.6980>, 2017.

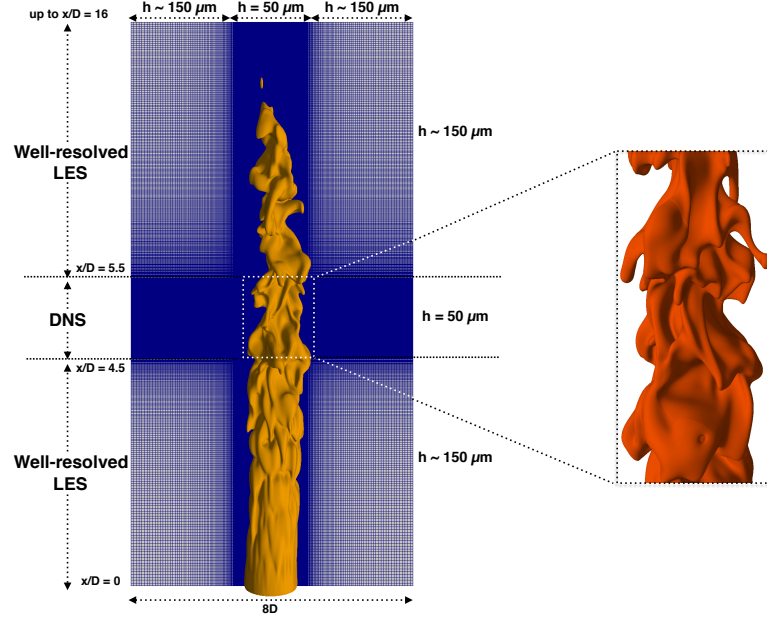


Figure 1: LES-DNS snapshot of the jet-flame simulation [41]. Mesh and iso-progress variable  $c = 0.8$ .  $h$ : resolution. Red iso-surface: zoom of iso- $c = 0.8$  in the DNS zone (different angle view).

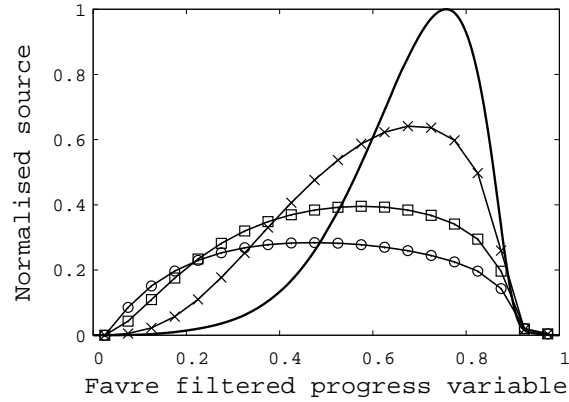
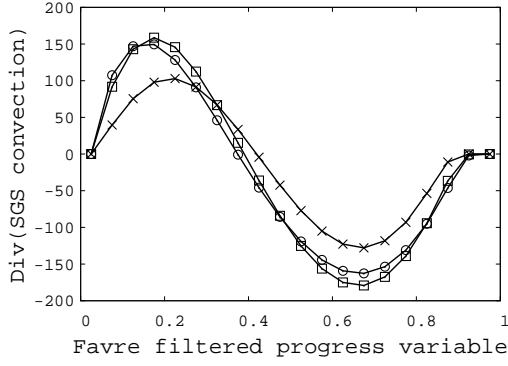
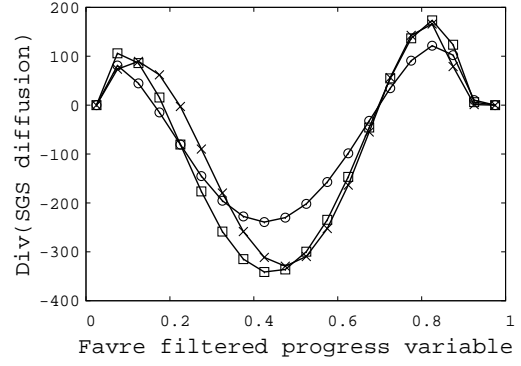


Figure 2: Thick-line:  $\bar{\omega}^+(\tilde{c})$  vs  $\tilde{c}$  as obtained from a 1D laminar flame (tabulated chemistry).  $\langle \bar{\omega}^+ | \tilde{c} \rangle$  from DNS vs  $\tilde{c}$  for filter sizes  $\times$ : 0.3 mm,  $\square$ : 0.6 mm,  $\circ$ : 0.9 mm ( $\delta_L = 0.4$  mm).

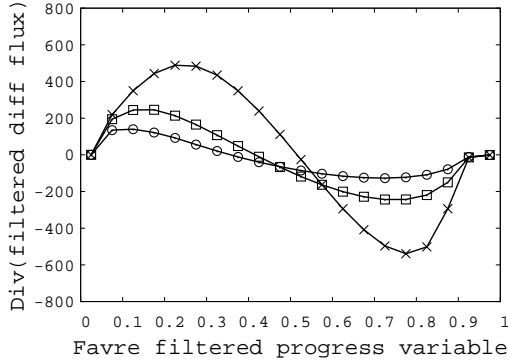




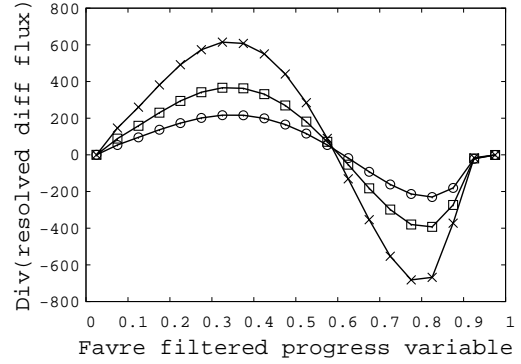
(a)  $\langle \nabla \cdot \tau_c | \tilde{c} \rangle$



(b)  $\langle \nabla \cdot \tau_D | \tilde{c} \rangle$



(c)  $\langle \overline{\nabla \cdot (\rho D_c(c) \nabla c)} | \tilde{c} \rangle$



(d)  $\langle \nabla \cdot (\bar{\rho} D_c(\tilde{c}) \nabla \tilde{c}) | \tilde{c} \rangle$

Figure 3: Conditional statistical means vs filtered progress variable. (a): Divergence of SGS convective scalar flux. (b): divergence of SGS diffusive flux. (c): filtered diffusive flux. (d): diffusive flux computed from the resolved quantities. Filter size  $\times$ : 0.3 mm,  $\square$ : 0.6 mm,  $\circ$ : 0.9 mm.

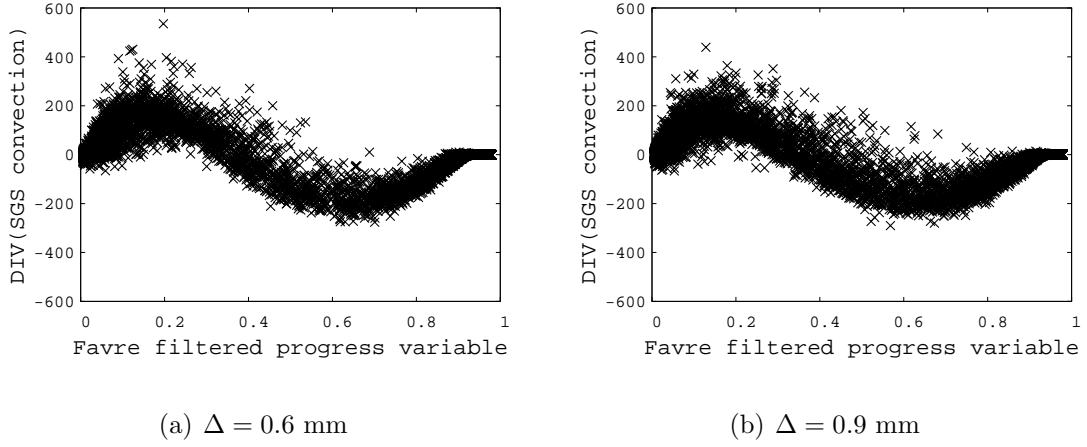


Figure 4: Scatter plot of  $\nabla \cdot \tau_c$  (1 every 100 DNS points shown).

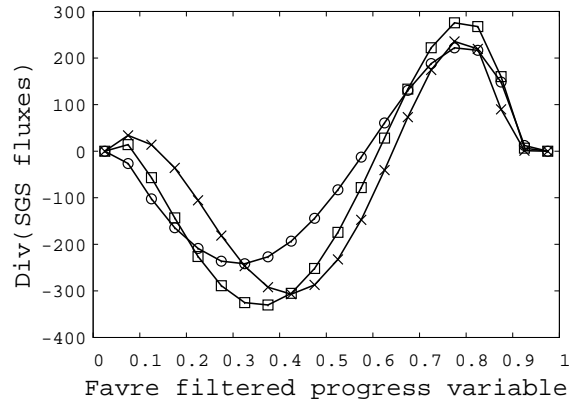


Figure 5:  $\langle \nabla \cdot \tau \mid \tilde{c} \rangle = \langle \nabla \cdot (\tau_D - \tau_c) \mid \tilde{c} \rangle$  vs  $\tilde{c}$ . Filter size  $\times$ : 0.3 mm,  $\square$ : 0.6 mm,  $\circ$ : 0.9 mm.

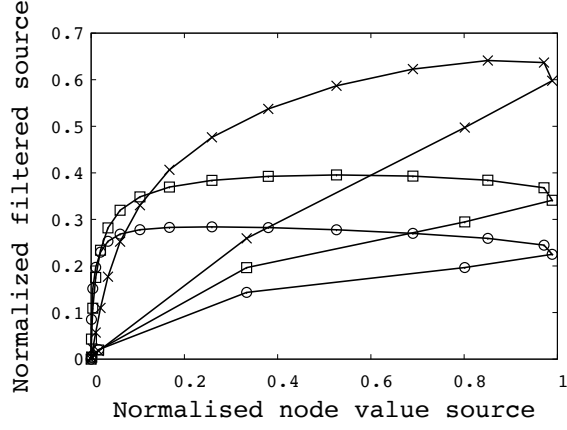


Figure 6:  $\langle \bar{\omega}^+(c) | \tilde{c} \rangle$  vs  $\dot{\omega}^+(\tilde{c})$ . Filter size  $\times$ : 0.3 mm,  $\square$ : 0.6 mm,  $\circ$ : 0.9 mm.

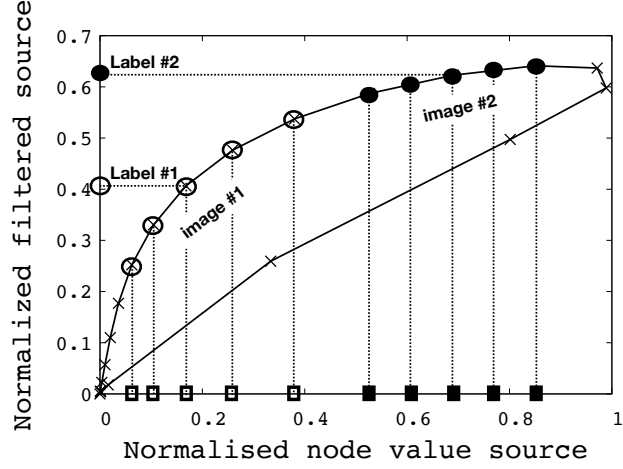
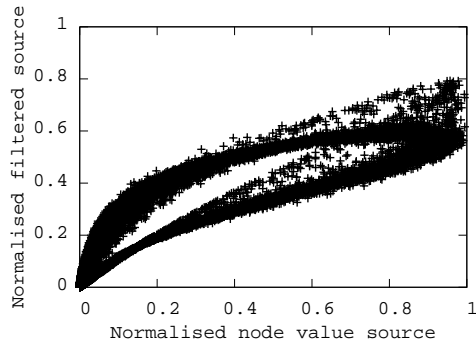
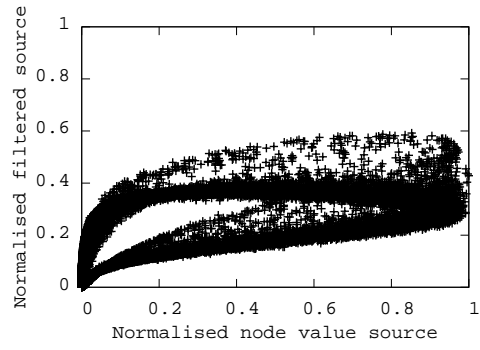


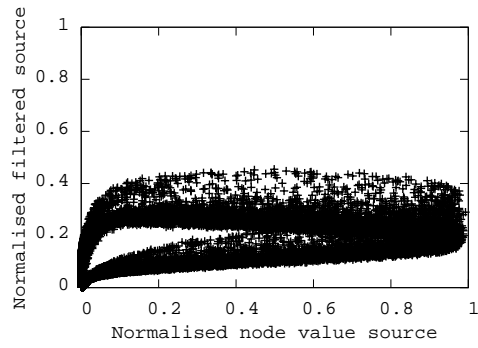
Figure 7:  $\langle \bar{\omega}^+(c) | \tilde{c} \rangle$  vs  $\dot{\omega}^+(\tilde{c})$ . Sketch of the construction of images and labels for training a CNN. Filter size: 0.3 mm.



(a)  $\Delta = 0.3$  mm

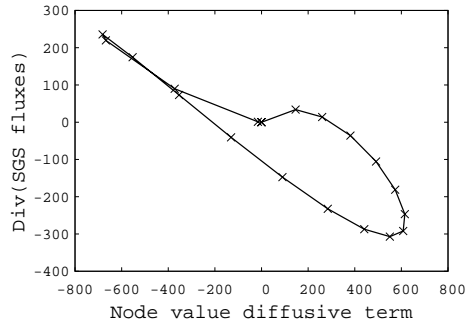


(b)  $\Delta = 0.6$  mm

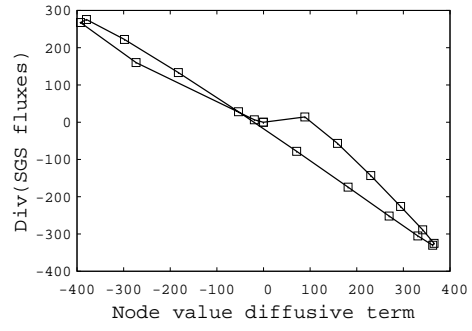


(c)  $\Delta = 0.9$  mm

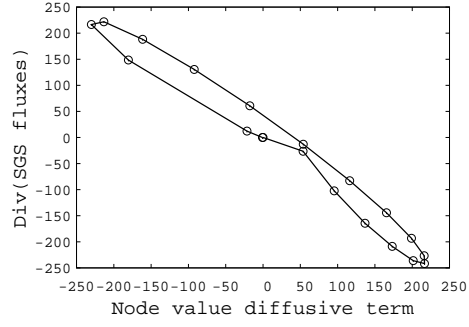
Figure 8:  $\overline{\dot{\omega}^+}$  vs  $\dot{\omega}^+(\tilde{c})$ .



(a)  $\Delta = 0.3$  mm



(b)  $\Delta = 0.6$  mm



(c)  $\Delta = 0.9$  mm

Figure 9:  $\langle \nabla \cdot \tau \mid \tilde{c} \rangle$  vs  $\langle \nabla \cdot (\bar{\rho} D_c(\tilde{c}) \nabla \tilde{c}) \mid \tilde{c} \rangle$ .

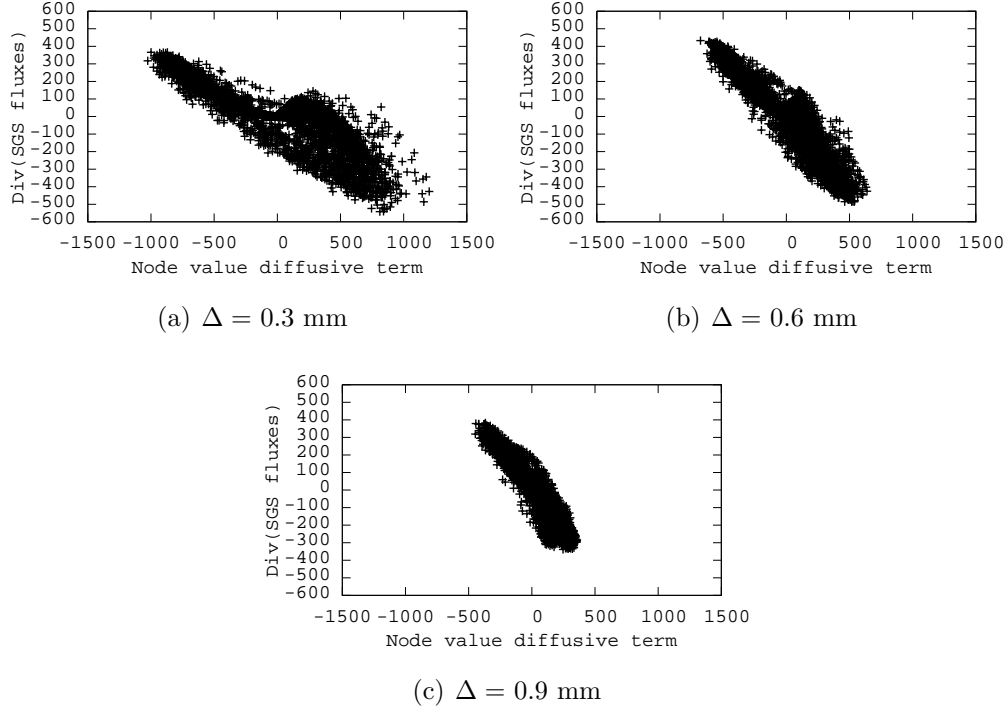


Figure 10:  $\nabla \cdot \tau$  vs  $\nabla \cdot (\bar{\rho} D_c(\tilde{c}) \nabla \tilde{c})$ .

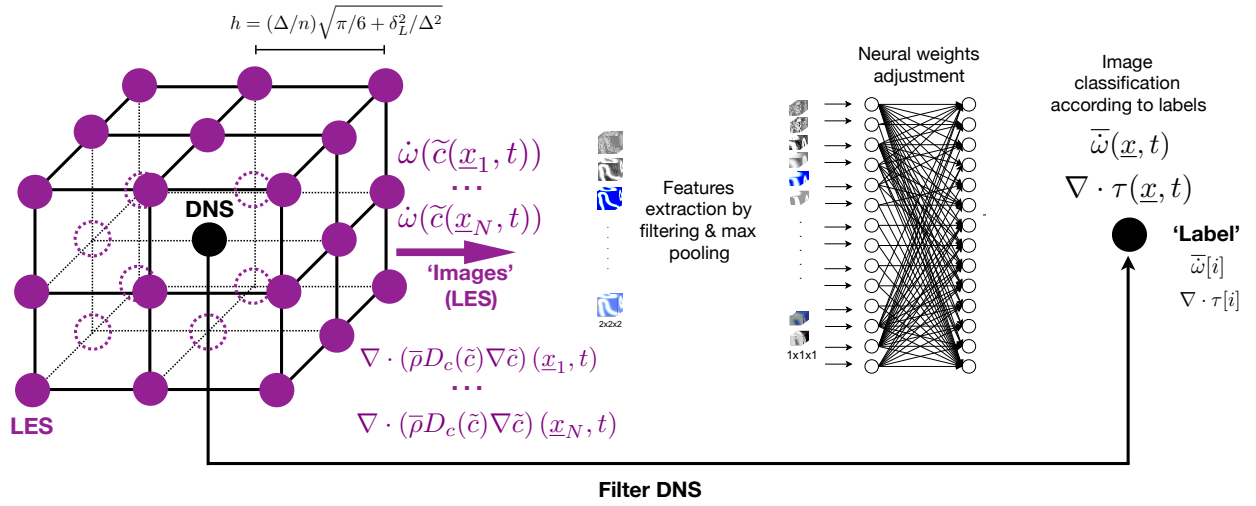


Figure 11: CNN training from DNS, sketch of the database construction.

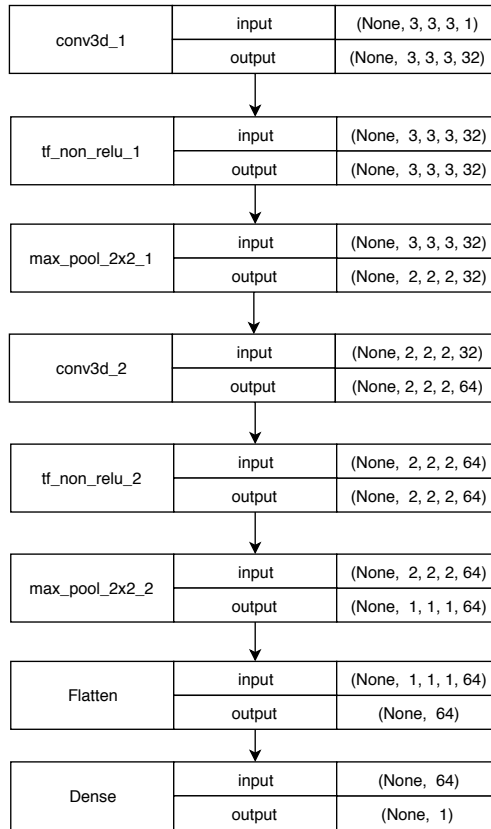
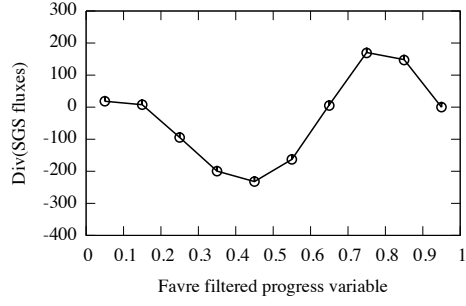
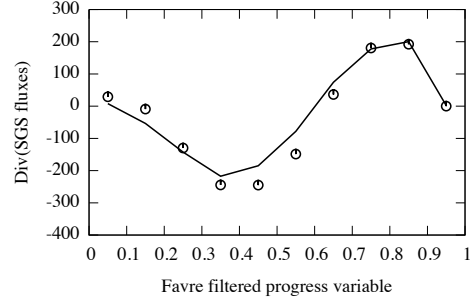


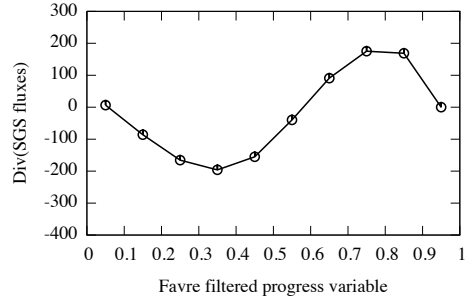
Figure 12: Structure of the convolutional neural network used (set of TensorFlow routines).



(a)  $\Delta = 0.30$  mm (trained)



(b)  $\Delta = 0.45$  mm (untrained)



(c)  $\Delta = 0.90$  mm (trained)

Figure 13:  $\langle \nabla \cdot \tau \mid \tilde{c} \rangle$  vs  $\tilde{c}$ . Symbols: DNS reference. Line: CNN prediction.



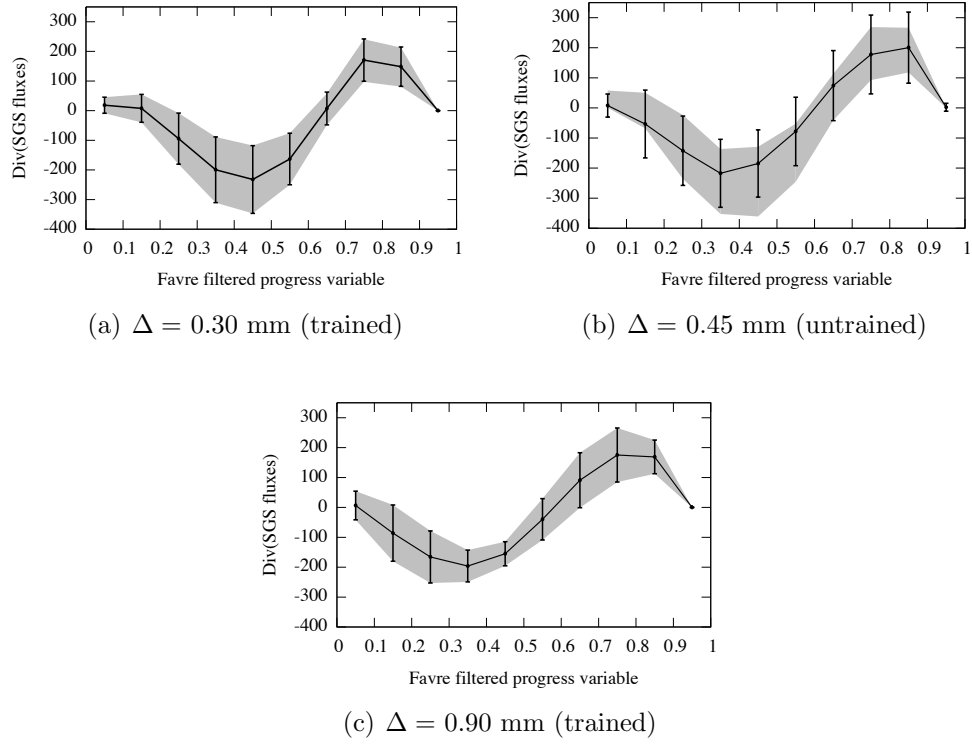
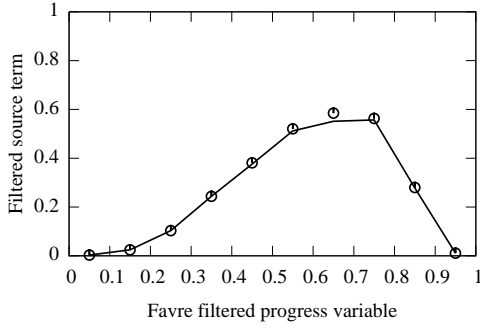
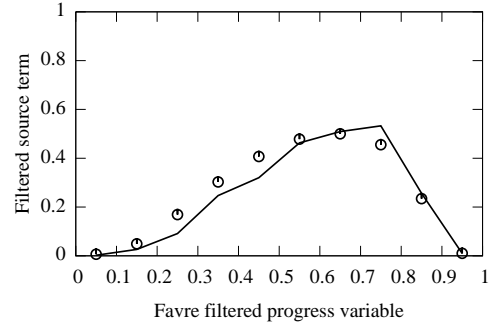


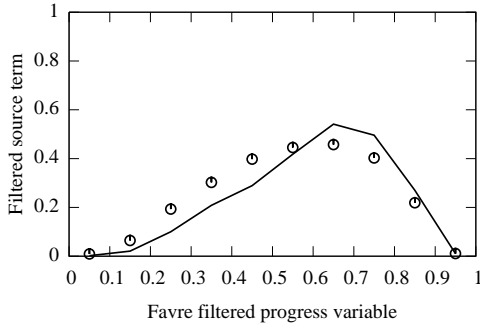
Figure 14: Solid line:  $\langle \nabla \cdot \tau \mid \tilde{c} \rangle$  vs  $\tilde{c}$  from CNN. Gray: Range covered by the signal according to the RMS in DNS, vertical bar: CNN prediction.



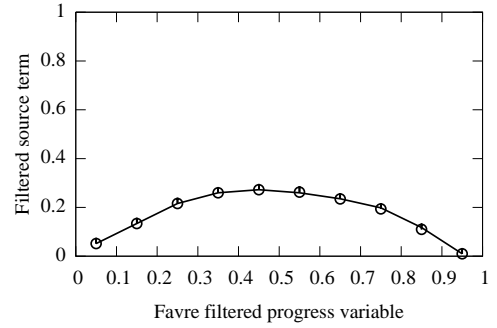
(a) Trained database.  $\Delta = 0.30$  mm



(b) Untrained database.  $\Delta = 0.40$  mm

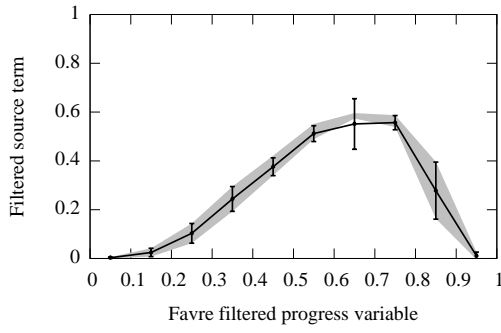


(c) Untrained database.  $\Delta = 0.45$  mm

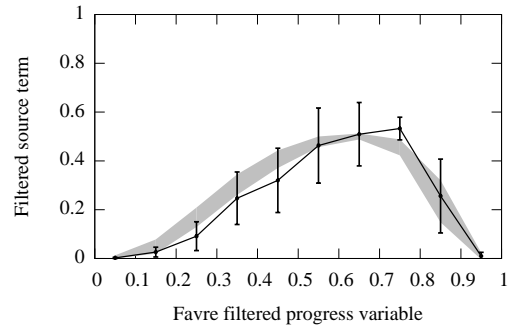


(d) Trained database.  $\Delta = 0.90$  mm

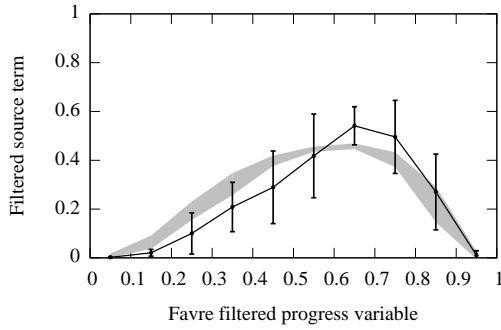
Figure 15:  $\langle \overline{\dot{\omega}^+} | \tilde{c} \rangle$  vs  $\tilde{c}$ . Symbols: DNS reference. Line: CNN prediction.



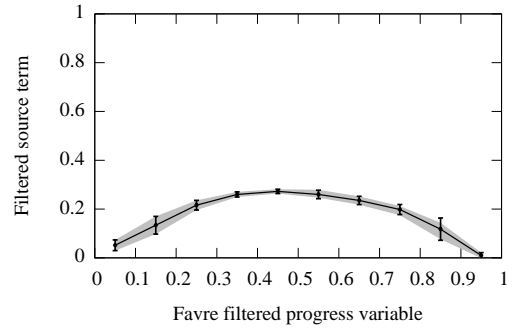
(a) Trained database.  $\Delta = 0.30$  mm



(b) Untrained database.  $\Delta = 0.40$  mm



(c) Untrained database.  $\Delta = 0.45$  mm



(d) Trained database.  $\Delta = 0.90$  mm

Figure 16: Solid line:  $\langle \bar{\omega}^+ | \tilde{c} \rangle$  vs  $\tilde{c}$  from CNN. Gray: Range covered by the signal according to the RMS in DNS, vertical bar: CNN prediction.